Time Integration and Assessment of a Model for Shape Memory Alloys Considering Multiaxial Nonproportional Loading Cases

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ABSTRACT
The paper presents a numerical implementation of the ZM model for shape memory alloys that fully accounts for non-proportional loading and its influence on martensite reorientation and phase transformation. Derivation of the time-discrete implicit integration algorithm is provided. The algorithm is used for finite element simulations using Abaqus, in which the model is implemented by means of a user material subroutine. The simulations are shown to agree with experimental and numerical simulation data taken from the literature.

1. INTRODUCTION
Shape memory alloys (SMAs) are capable of undergoing severe inelastic deformation that can be recovered by heating. Alternatively, beyond a certain temperature, the recovery of inelastic deformation can be accomplished by removal of the load to which the SMA is subjected. Such behavior is known as “superelasticity” because it holds phenomenological analogy with conventional elasticity, even though the underlying process is dissipative and gives rise to a hysteresis loop in the SMA stress-strain curve.

The present work focuses on proper numerical integration of the Zaki-Moumni model for SMAs subjected to complex multiaxial loading in the superelastic range. The inelastic deformation of the material is accomplished in this case by means of two distinct physical processes: a solid-solid phase transformation from a single-variant parent phase called austenite to a multivariant product phase called martensite, and a reorientation of martensite by the formation of preferred variants at the expense of others as a result of changes in the magnitude and orientation of the stress experienced by the martensite. Other aspects of SMA behavior such as tensile-compressive asymmetry, SMA dynamics, slip plasticity and thermomechanical coupling including cyclic effects are not considered here. The time integration of the Zaki-Moumni model for martensitic SMAs is addressed in [15, 16].

In the literature, several models for SMAs were proposed that account for multiaxial loading to various degrees of success. [4] developed a model for SMAs that was validated against experimental data obtained earlier by the same research group for samples subjected to nonproportional biaxial loading. [11] derived a phenomenological model where the state equations are derived from thermo-
dynamic potentials in accordance with the principles of thermodynamics. The model was validated against experimental data obtained by [12]. Other models were also presented in [13], [2], and more recently in [1]. The latter uses the Fischer-Burgmeister functions to bold the Kuhn-Tucker optimality conditions with nonlinear equalities in order to avoid the need for iterative detection of active loading surfaces.

Simulation results using the ZM model for SMAs are compared in this paper to some of those reported in the above references. The model is further validated using experimental data taken from the literature.

2. ANALYTICAL DERIVATIONS

Only key constitutive relations are listed here for conciseness. For details on the analytical derivation procedure for the model the reader is referred to [17, 18]. The following state variables are considered: the volume fraction of martensite $z$, the local inelastic strain in the martensite phase due to reorientation $\varepsilon^{ori}$, as well as the conventional total strain $\varepsilon$ and temperature $T$. Following standard convention, tensors are represented with bold characters.

The ZM model uses the framework of generalized standard materials in deriving the constitutive relations for SMAs. The state equations are derived from the expression of a Helmholtz free energy density and include the following stress-strain relation:

$$\sigma = K : (\varepsilon - \varepsilon^{ori}),$$

where $\sigma$ is the stress tensor and $K$ is the elastic stiffness tensor, which may depend on the volume fraction of martensite.

The above equation reflects that the total inelastic strain in a reference volume element of the SMA is resolved into the product of the amount of martensite that exists within that element and the amount of local inelastic strain experienced by the martensite due to reorientation of the variants.

Both $z$ and $\varepsilon^{ori}$ are dissipative variables for which the evolution is governed by the loading functions $\mathcal{F}_z^1$ and $\mathcal{F}_z^2$ for forward $(\dot{z} > 0)$ and reverse $(\dot{z} < 0)$ phase transformations and $\mathcal{F}_{ori}$ for martensite reorientation. The loading functions depend on $\mathbf{\sigma}$, $z$, $\varepsilon^{ori}$ and $T$. For the purpose of this manuscript, only isothermal processes are considered, temperature is therefore introduced only as a parameter.

The explicit expressions for the loading functions are the following:

$$\mathcal{F}_z^1 = F_z^1 (\mathbf{\sigma}, z, \varepsilon^{ori})$$

$$= \frac{1}{2} \left( 3 E_{\text{MA}} s + P_{\text{MAT}}^2 (\mathbf{\sigma}) \right) - C(T)$$

$$+ s : \varepsilon^{ori} - (G + b)z - a(1 - z)$$

$$- \left[ (\alpha - \beta)z + \frac{\beta}{2} \right] \left( \frac{2}{3} \varepsilon^{ori} : \varepsilon^{ori} \right),$$  \hfill (2)

$$\mathcal{F}_z^2 = F_z^2 (\mathbf{\sigma}, z, \varepsilon^{ori})$$

$$= - \frac{1}{2} \left( \frac{3}{2} E_{\text{MA}} s + P_{\text{MAT}}^2 (\mathbf{\sigma}) \right) + C(T)$$

$$- s : \varepsilon^{ori} - (G - b)z - a(1 - z)$$

$$+ \left[ (\alpha + \beta)z + \frac{\beta}{2} \right] \left( \frac{2}{3} \varepsilon^{ori} : \varepsilon^{ori} \right),$$  \hfill (3)

$$\mathcal{F}_{ori} = F_{ori} (\mathbf{\sigma}, z, \varepsilon^{ori}) = X_{vm} - \zeta Y$$  \hfill (4)

with

$$X = s - \frac{2}{3\epsilon_0} (s : \varepsilon^{ori}) \varepsilon^{ori},$$  \hfill (5)

where $s$ is the stress deviator, $\text{tr}(\mathbf{\sigma})$ is the trace of the stress tensor, $X_{vm}$ is the von Mises equivalent of the thermodynamic force $X$, $C(T)$ is a linear function of temperature and $E_{\text{MA}}, P_{\text{MAT}}, \alpha, \beta, a, b, G$ are material parameters. The evolution of $\varepsilon^{ori}$ is governed by the normality rule

$$\dot{\varepsilon}^{ori} = \eta \frac{\partial \mathcal{F}_{ori}}{\partial X} = \frac{3}{2} \eta \frac{X}{X_{vm}} = \eta \mathbf{N}$$  \hfill (6)

where $\eta$ is an inelastic multiplier and $\mathbf{N} = \frac{2}{3} \frac{X}{X_{vm}}$ is the direction of the orientation strain rate $\varepsilon^{ori}$ in strain space. The inelastic multiplier $\eta$ and the rate of phase transformation $\dot{z}$ are governed by standard Kuhn-Tucker conditions.

3. ALGORITHMIC SETUP

The problem to be solved is that of a SMA structure subjected to arbitrary mechanical loading over a time interval $[0, T]$. Following standard incremental solution procedure, the time interval is discretized into $N$ subintervals. Starting from a well-defined initial state, the time-discrete model used here are

where $\Delta$ is the change of the increment of stress in terms of the increment of strain.

Assuming strain-controlled time-integration and using the symbol $\Delta$ to indicate time-discrete increments, the time-discrete equations for the SMA model used here are written as follows:
1. Elastic predictor:

Set $k = 0,$

$$\epsilon_{n+1}^{(k)} = \epsilon_n + \Delta \epsilon$$  \hspace{1cm} (7)

$$z_{n+1} = z_n$$  \hspace{1cm} (9)

$$\epsilon_{n+1}^{\text{ori}} = \epsilon_n^{\text{ori}}$$  \hspace{1cm} (10)

$$\sigma^{(k)}_{n+1} = K_n \cdot \left( \epsilon_n^{(k)} - z_{n+1} \epsilon_{n+1}^{\text{ori}} \right)$$  \hspace{1cm} (11)

2. Consistency conditions:

(a) Loading functions

$$F_z^{(1)}(k) = F_1 \left( \sigma^{(k)}, z^{(k)}, \epsilon^{\text{ori}}(k) \right)$$  \hspace{1cm} (12)

$$F_z^{(2)}(k) = F_2 \left( \sigma^{(k)}, z^{(k)}, \epsilon^{\text{ori}}(k) \right)$$  \hspace{1cm} (13)

$$F_{\text{ori}}(k) = F_{\text{ori}} \left( \sigma^{(k)}, z^{(k)}, \epsilon^{\text{ori}}(k) \right)$$  \hspace{1cm} (14)

(b) Active loading set

If $z^{(k)} < 1$ and $F_z^{(1)}(k) > 0$ then forward phase change,

If $z^{(k)} > 0$ and $F_z^{(2)}(k) > 0$ then reverse phase change,

If $F_{\text{ori}}(k) > 0$ then martensite reorientation.

(c) Increments of internal variables

If forward phase change and no martensite reorientation then solve $F_z^{(1)}(k+1) = 0$ for $\Delta z^{(k+1)}, \Delta \eta^{(k+1)} = 0,$

If reverse phase change and no martensite reorientation, solve $F_z^{(2)}(k+1) = 0$ for $\Delta z^{(k+1)}, \Delta \eta^{(k+1)} = 0,$

If martensite reorientation and no phase transformation then solve $F_{\text{ori}}^{(k+1)} = 0$ for $\Delta \eta^{(k+1)}, \Delta z^{(k+1)} = 0,$

If forward phase change and martensite reorientation then solve the system \[ \{ F_z^{(1)}(k+1) = 0, F_{\text{ori}}^{(k+1)} = 0 \} \] for $\Delta z^{(k+1)}$ and $\Delta \eta^{(k+1)},$

If reverse phase change and martensite reorientation then solve the system \[ \{ F_z^{(2)}(k+1) = 0, F_{\text{ori}}^{(k+1)} = 0 \} \] for $\Delta z^{(k+1)}$ and $\Delta \eta^{(k+1)}.$

(d) Positivity of multipliers

If forward phase change and $\Delta z^{(k+1)} > 0$ then set $\Delta z^{(k+1)} = 0,$ forward phase change is inactive,

If reverse phase change and $-\Delta z^{(k+1)} < 0$ then set $\Delta z^{(k+1)} = 0,$ reverse phase change is inactive,

If martensite reorientation and $\Delta \eta^{(k+1)} > 0$ then set $\Delta \eta^{(k+1)} = 0,$ martensite reorientation is inactive.

(e) Consistency with intrinsic constraints

If forward phase change and $z_n + \Delta z^{(k+1)} > 1$ then $\Delta z^{(k+1)} = 1 - z_n.$

If forward phase change and $z_n + \Delta z^{(k+1)} < 0$ then $\Delta z^{(k+1)} = -z_n.$

(f) Set $k = k + 1,$ update $\sigma^{(k)}, z^{(k)}, \epsilon^{\text{ori}}(k), N^{(k)}$ and repeat steps b to e until consistency with the loading conditions and the intrinsic constraints on $z$ and $\epsilon^{\text{ori}}$ is achieved.

3. Internal variables and stress update

$$z_{n+1} = z_n + \Delta z,$$  \hspace{1cm} (15)

$$\epsilon_{n+1}^{\text{ori}} = \epsilon_n^{\text{ori}} + \Delta \eta N_{n+1},$$  \hspace{1cm} (16)

$$\sigma_{n+1}^{\text{trial}} = K_{n+1} \cdot (\epsilon_{n+1} - z_{n+1} \epsilon_{n+1}^{\text{ori}}).$$  \hspace{1cm} (17)

In the above procedure, the equations $\mathcal{F} = 0,$ where $\mathcal{F} = F(\sigma, z, \epsilon^{\text{ori}})$ and $F$ is any of the loading functions, can be solved at iteration $k + 1$ using a Newton-Raphson algorithm such that

$$\mathcal{F}^{(k+1)} \approx \mathcal{F}^{(k)} + \mathcal{F}_\sigma^{(k)} : \Delta \sigma^{(k+1)} + \mathcal{F}_z^{(k)} : \Delta z^{(k+1)} + \mathcal{F}_{\text{ori}}^{(k)} : \Delta \epsilon^{\text{ori}}(k+1),$$  \hspace{1cm} (18)

where a comma used in the subscript indicates differentiation with respect to the subsequent variable. Introducing the approximate normality rule

$$\Delta \epsilon^{\text{ori}}(k+1) \approx \Delta \eta^{(k+1)} N^{(k)}$$  \hspace{1cm} (19)

where the direction tensor $N$ is approximated using its value at iteration $k,$ the above leads to a linear algebraic equation with two unknowns $\Delta z^{(k+1)}$ and $\Delta \eta^{(k+1)}.$ A system of two such equations is solved every time the increments of the internal variables are determined.

4. NUMERICAL SIMULATION AND VALIDATION

The model is used to simulate the experiment reported by Bouvet et al. in [3] for a SMA tube subjected to a combination of tension and internal pressure. The parameters of the model are determined using the experimental curve in figure 1 and the simulated curve is shown on the same figure for comparison. The obtained parameters are listed in table 1. In this table, $E_A$ and $E_{AB}$ and Young’s moduli for austenite and martensite, $v$ is Poisson’s coefficient for the SMA, $Y$ is the stress onset for martensite detwinning at low temperature, $\xi$ and $\kappa$ are parameters used to define the function $C(T),$ and $A_T^0$ is the austenite-finish transformation temperature at zero stress.

The loading to which the cylinder is subjected corresponds to the axial and hoop stresses reported in figure 2. The behavior of the material is reported in terms of hoop
Fig. 1: Experimental and fitted stress-strain curves for the experiment of Bouvet et al. in [3].

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<th>Parameter</th>
<th>Value</th>
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<td>$A_f^0$</td>
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Table 1: Parameters used for simulating the experiment of Bouvet et al.

vs axial strain in figure 3 and in terms of the stress-strain curve in the axial and hoop directions in figures 4 and 5. The simulation results are in good agreement with the experimental data for the first, second, and fourth loading steps. A marked deviation is observed however for the second loading step, in which the variation in hoop strain is significantly underestimated by the model. This may be explained by anisotropic material behavior in the axial and hoop directions that is not accounted for by the present model.

5. CONCLUSION

An integration procedure for a model for shape memory alloys was presented that accounts for complex non-proportional loading cases in the superelastic range. The numerical integration procedure for the model was presented, including the steps necessary for the detection of active loading sets and the enforcement of intrinsic and consistency constraints on the state variables. The approach used is analogous to classical multisurface plasticity. The model was successfully used to simulate experimental data taken from the literature for a SMA sample subjected to biaxial loading.

Fig. 2: Nonproportional loading considered for the simulation.

Fig. 3: Hoop vs axial strain in the SMA cylinder.

Fig. 4: Axial stress-strain response for the SMA cylinder.

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Fig. 5: Hoop stress-strain response for the SMA cylinder.

REFERENCES


